

# Flow Control by DBD Plasma Actuator

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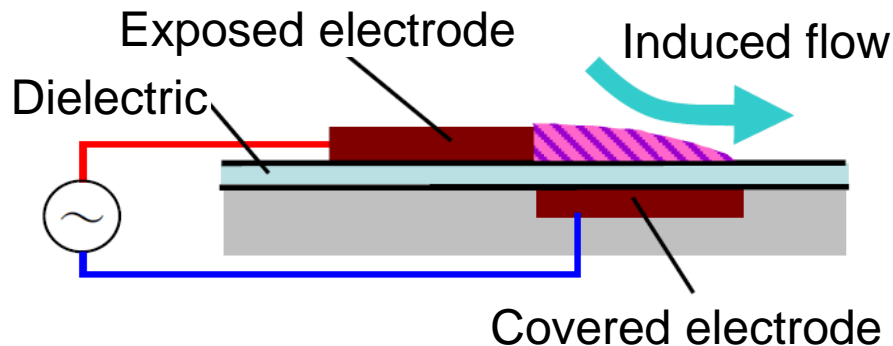
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# Dielectric-Barrier-Discharge (DBD) plasma actuator

## The popular device for flow control



High AC voltage

Dielectric barrier discharge

⇒ Plasma

⇒ Induced flow

### Advantage

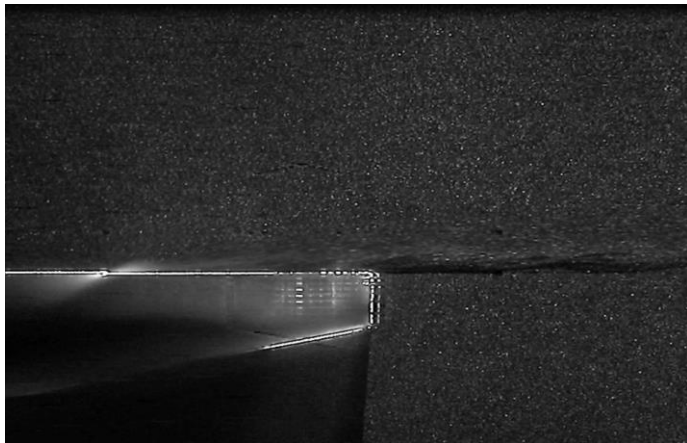
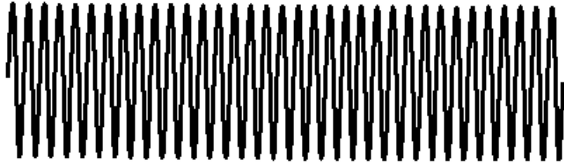
- High response
  - Simple structure
  - Low power consumption
- etc.

### Disadvantage

- Low momentum of the induced flow

# Waveforms of applied voltage (Pulse modulation drive)

Continuous drive



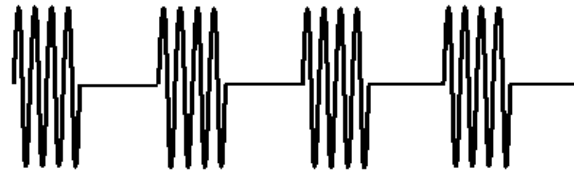
Continuous flow

Momentum is introduced to the flow.

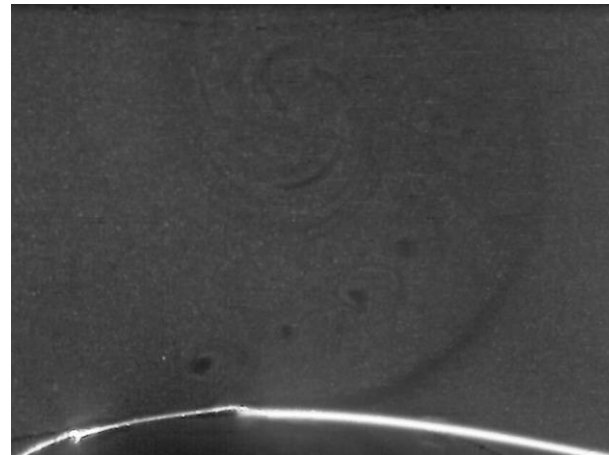
Not effective

(Low momentum of the induced flow)

Pulse modulation drive



in stationary air



Intermittent flow

Pulse modulation drive

Perturbation is introduced to the flow.

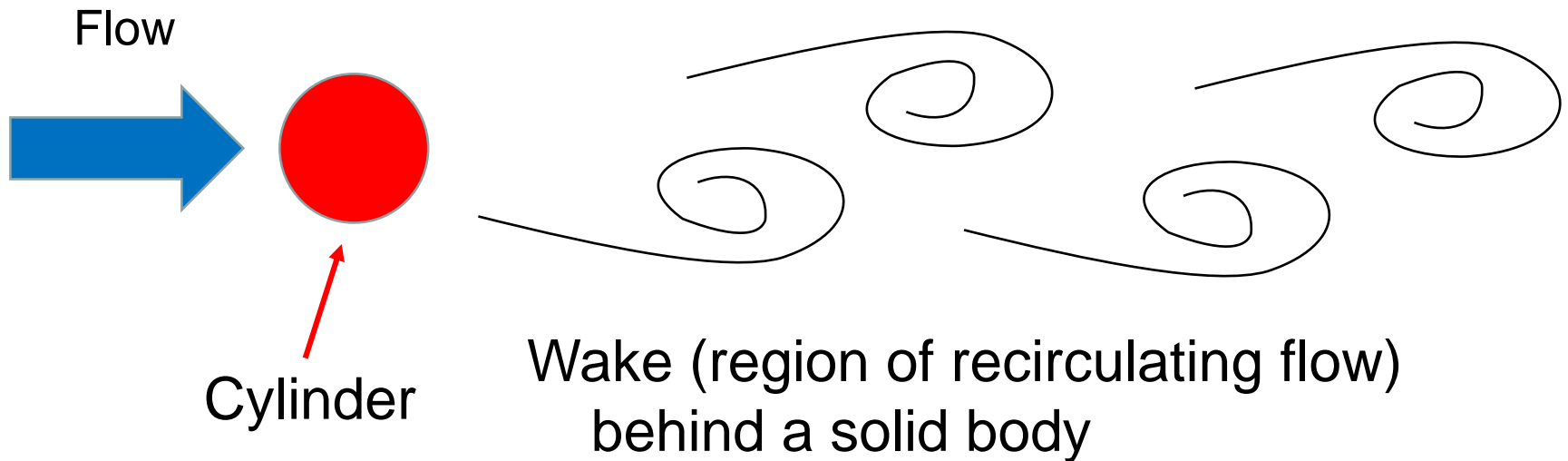
Timing control of flow phenomena is possible.

Effective for separation flow.

⇒ We use this control.

# 1<sup>st</sup> control example

## Wake of a cylinder



The wake of a cylinder :  
the staggered arrangement of vortices

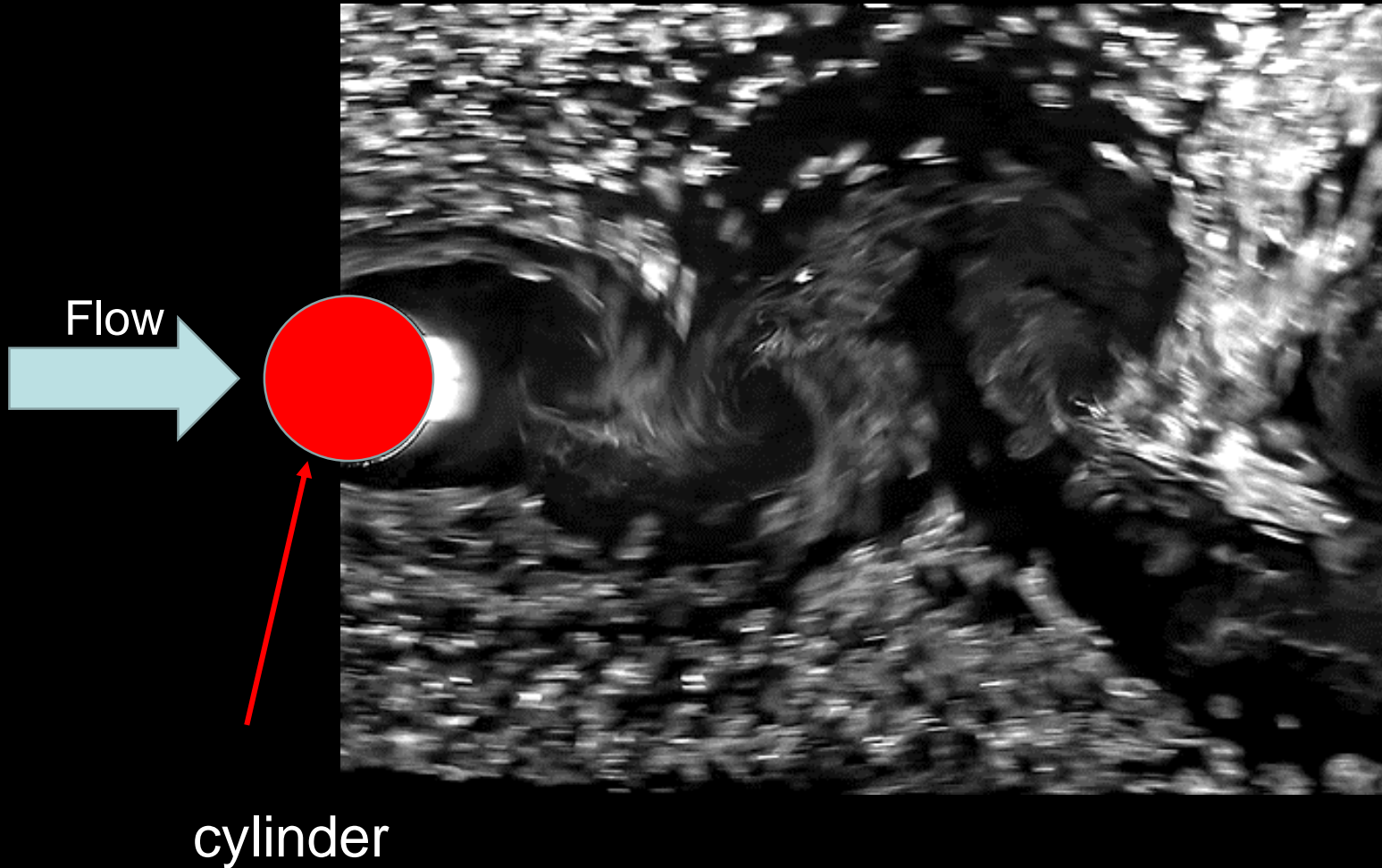
**Karman vortex street**

# Karman vortex street

Flow Visualization by smoke wire method

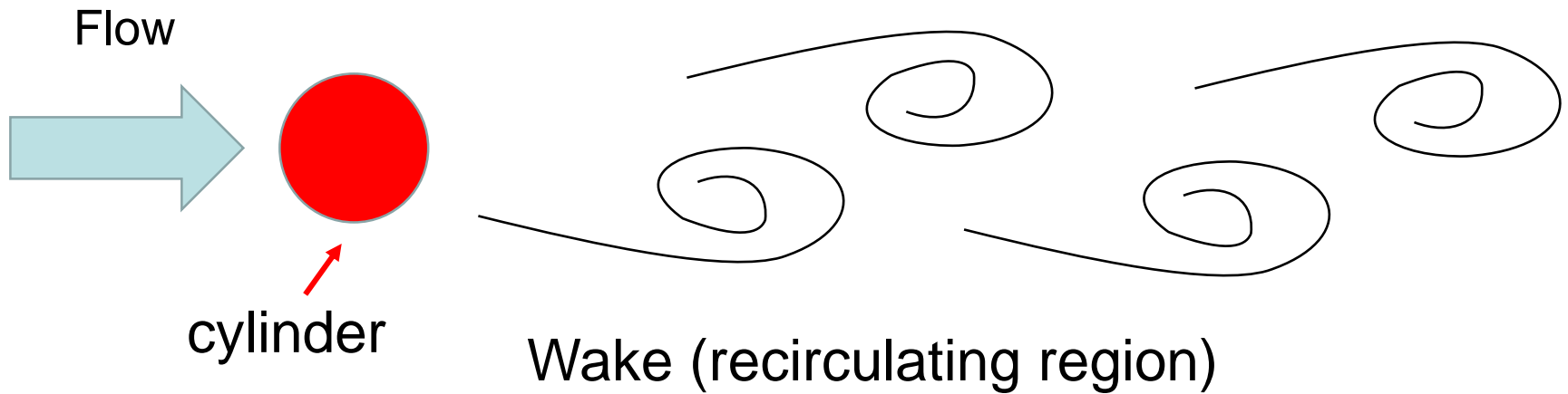
(play speed 1/67)

without control



# 1<sup>st</sup> control example

## Wake of a cylinder

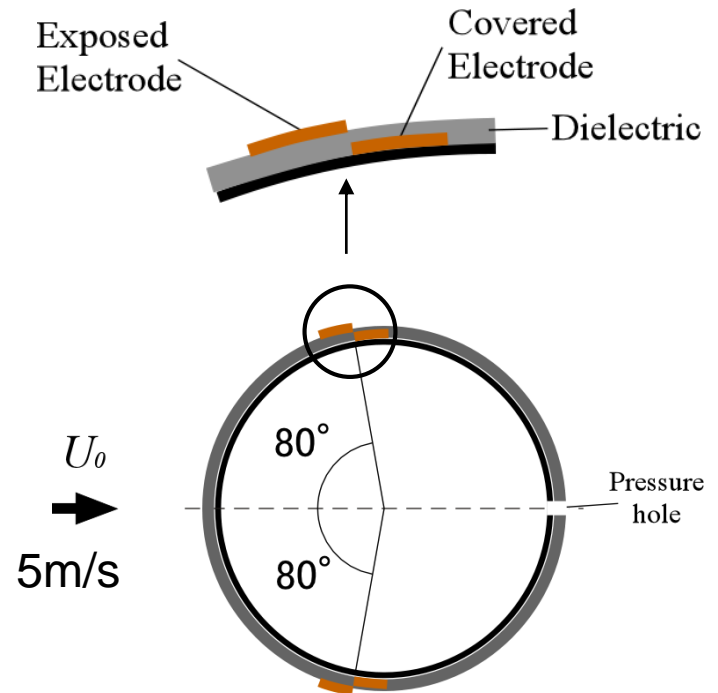
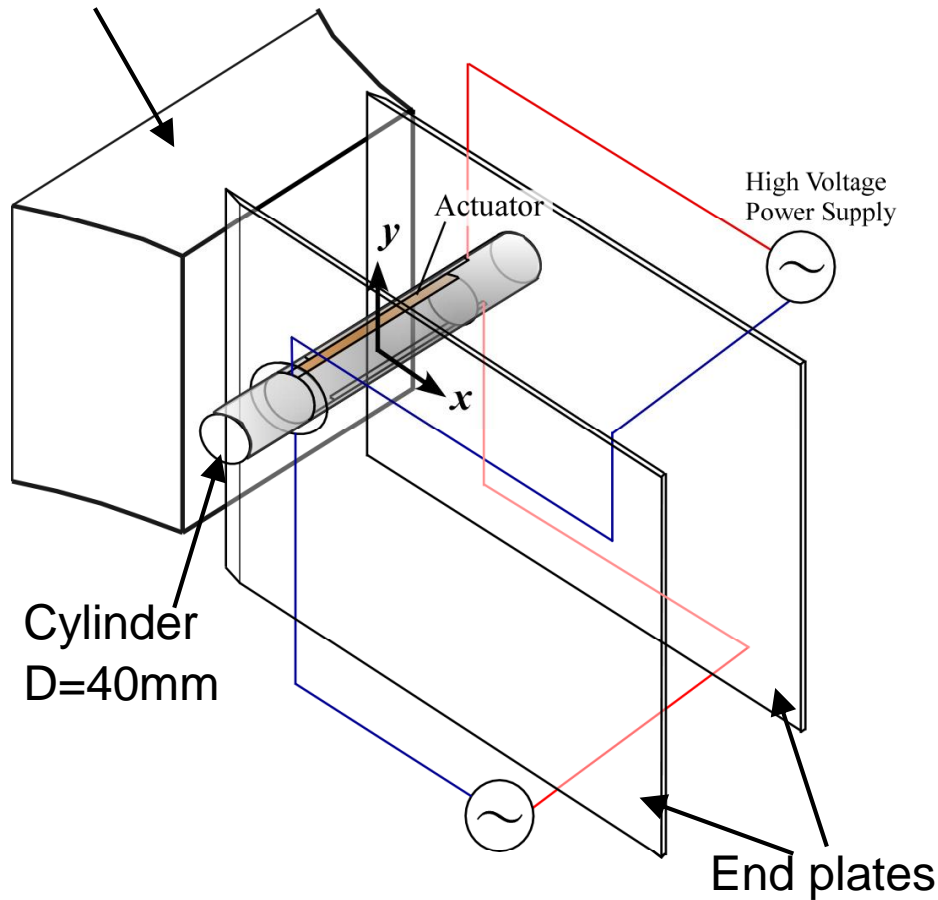


In Karman vortex street,  
the arrangement of vortices changes with time.  
⇒ Cause vibration and noise

Control the Karman vortex street

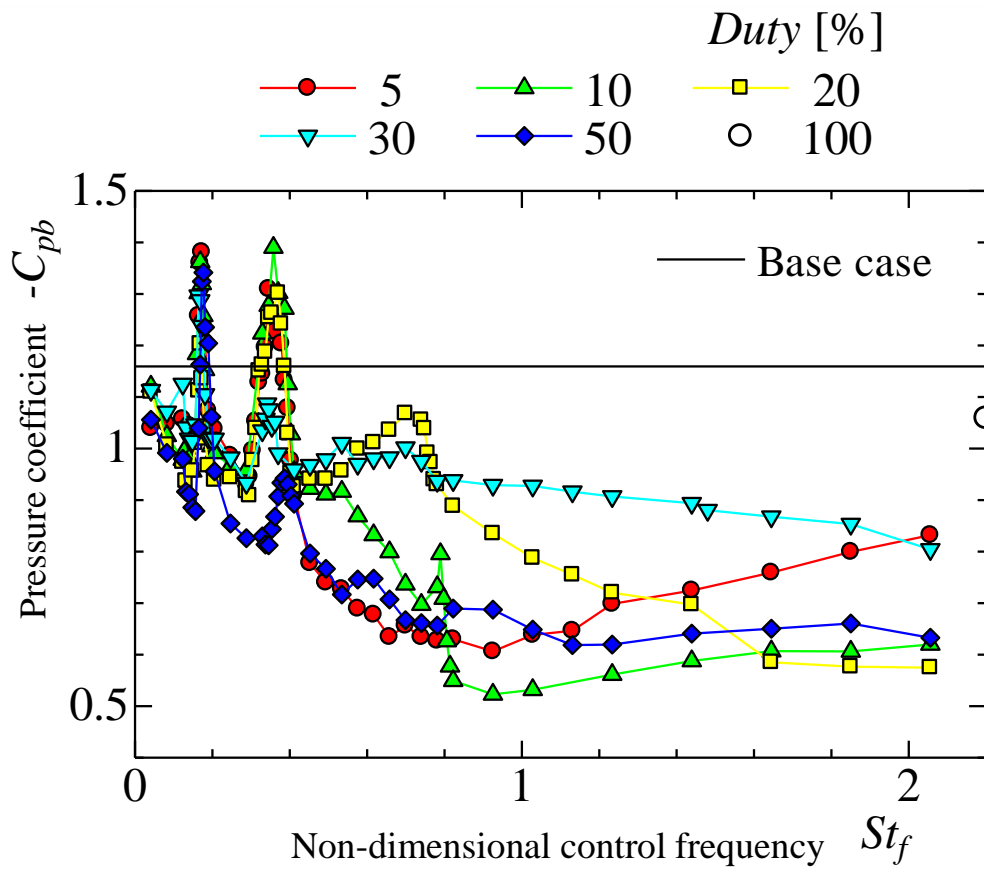
# Experimental setup

Nozzle of Wind tunnel



# Result

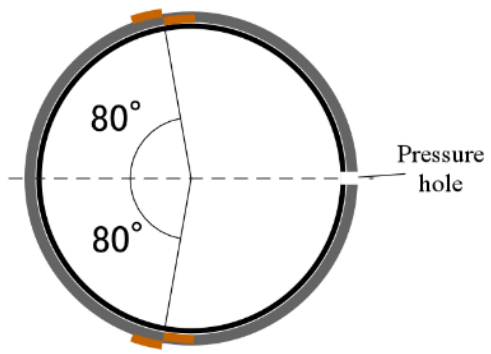
## Backpressure of cylinder



Drag  $U_0$  →

Increase

Decrease



Measurement position

This pressure give an indication of drag.

We can control the drag of cylinder.

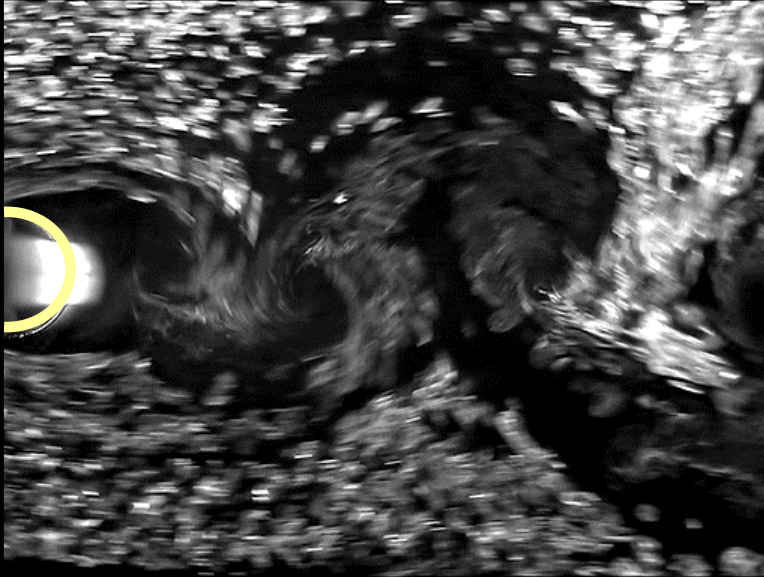
$$C_{pb} = \frac{2p_b}{\rho U_0^2} \quad St_f = \frac{f_M D}{U_0}$$



# Flow Visualization

(play speed 1/67)

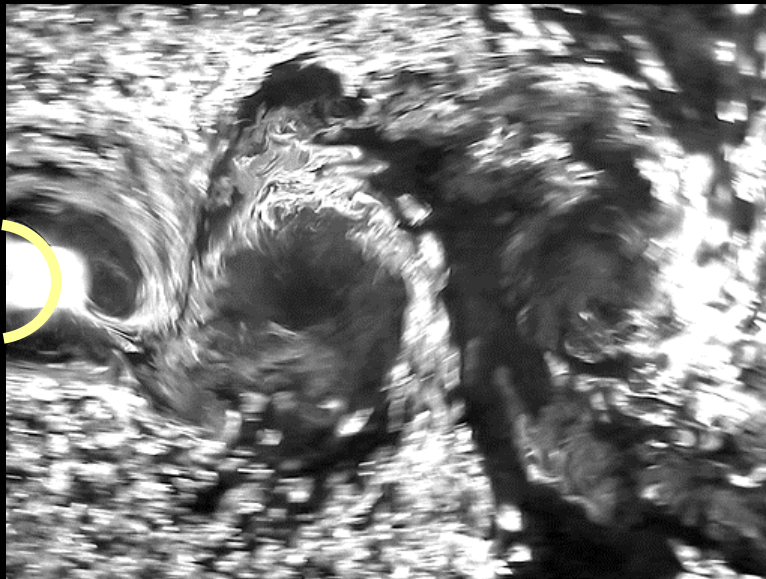
Base case



Drag decrease case



Drag increase case



Karman vortex street disappears.

Karman vortex street is remaining.

Karman vortex street can be controlled by the DBD plasma actuator.

# 2<sup>nd</sup> control example

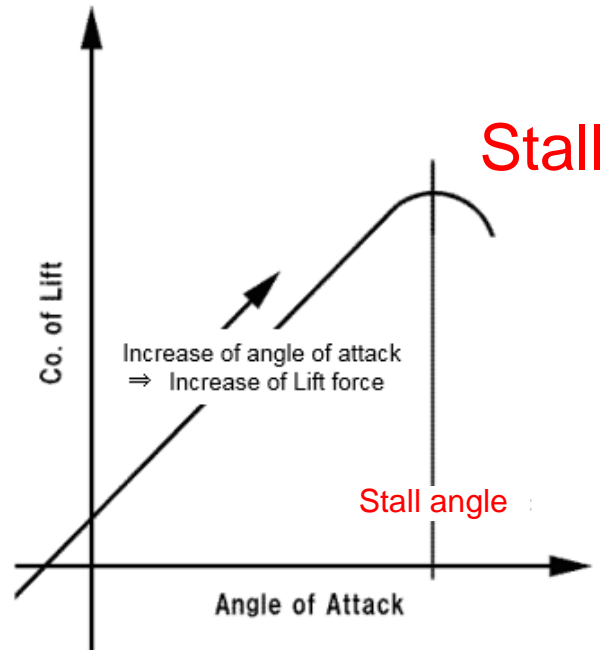
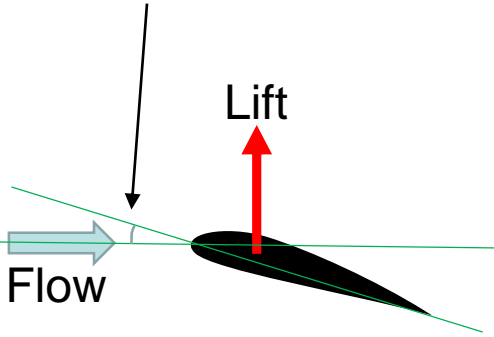
## Flow separation over a wing

Wing

Important parameter  
Angle of attack

Lift

Flow

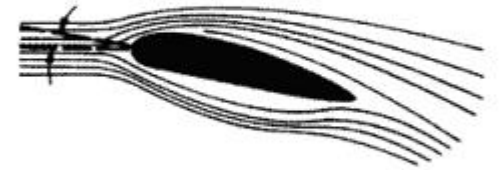


Change of lift by the angle of attack

LOW ANGLE OF ATTACK



MEDIUM ANGLE OF ATTACK



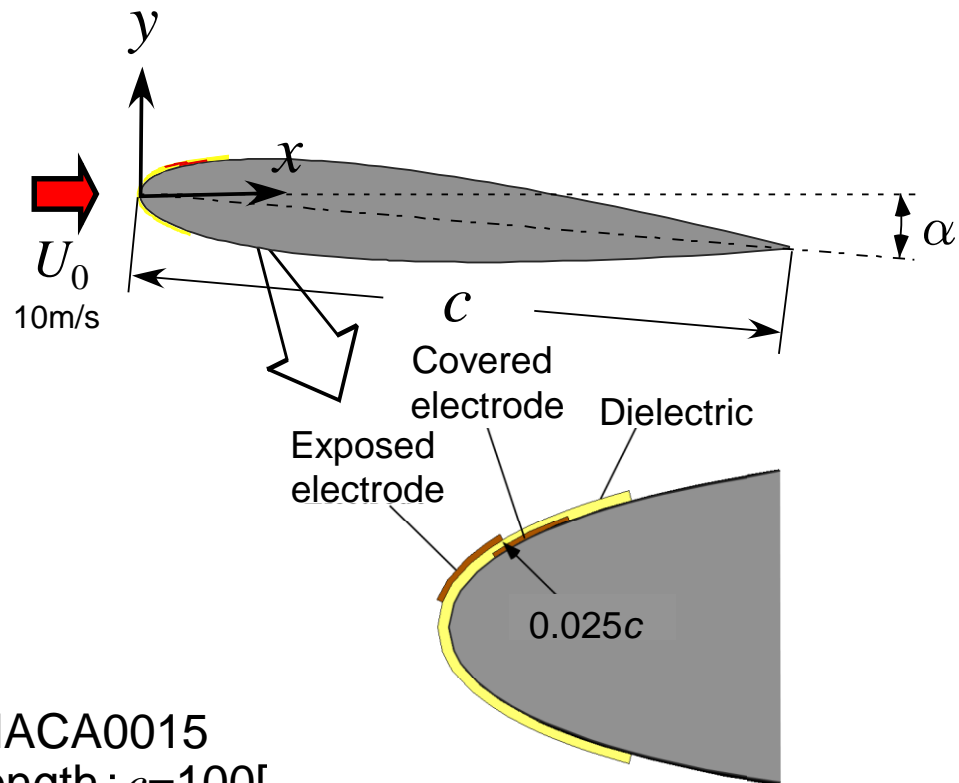
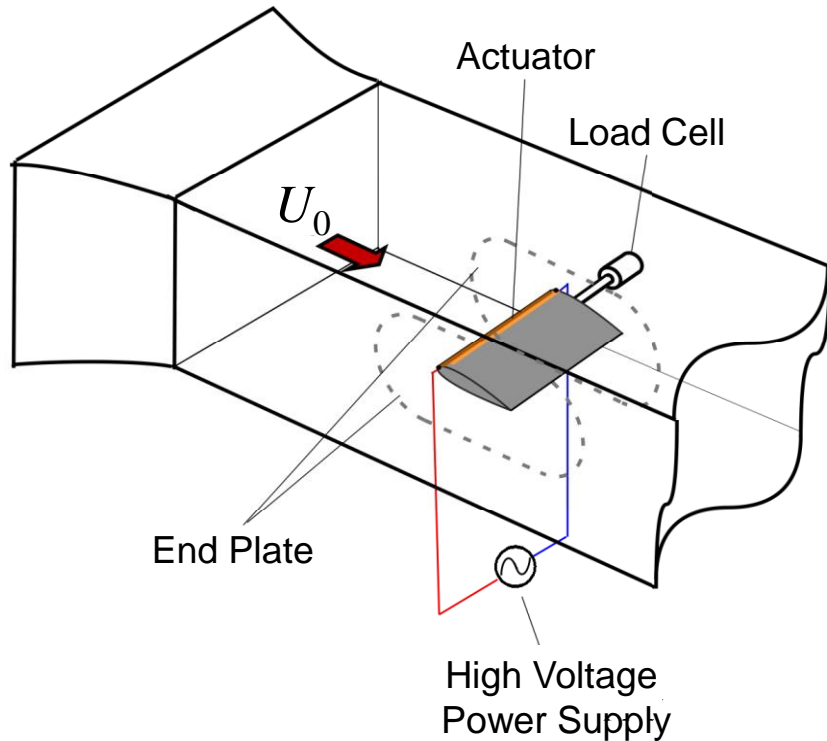
EXCESSIVE ANGLE OF ATTACK. AIR FLOW IS  
BROKEN. NO LIFT TO  
SUSTAIN THE  
AIRPLANE



The aim of this control is the prevention of separation.

⇒ Increase of lift force at high angle of attack.

# Experimental setup

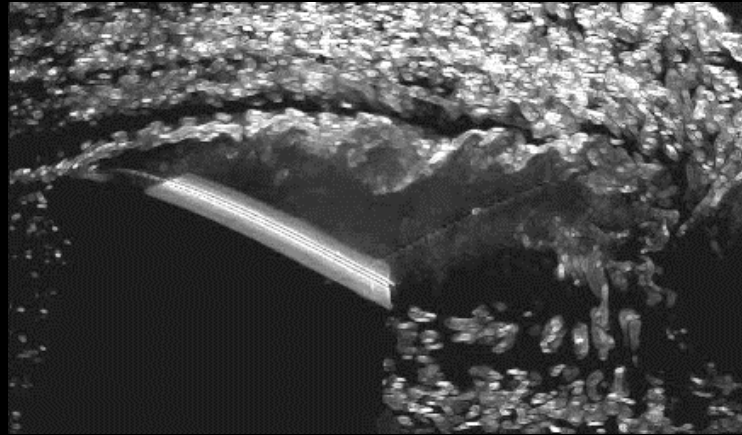


Airfoil : NACA0015  
Chord length :  $c=100$ [  
Span length :  $s=150$ [mm]  
Aspect ratio :  $AR=1.5$

# Flow visualization

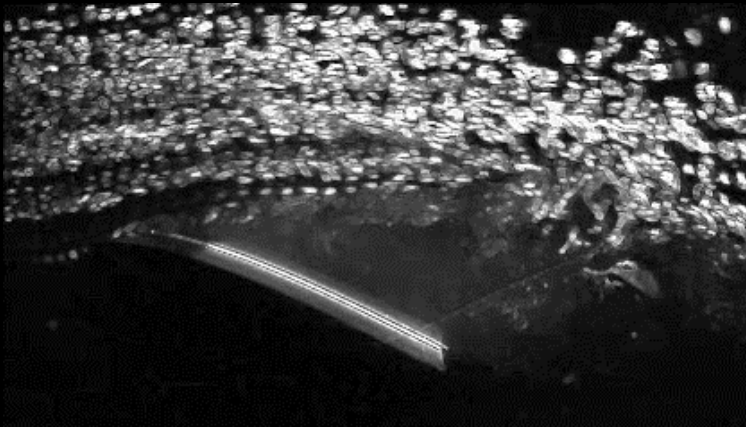
(high angle of attack)

No actuation

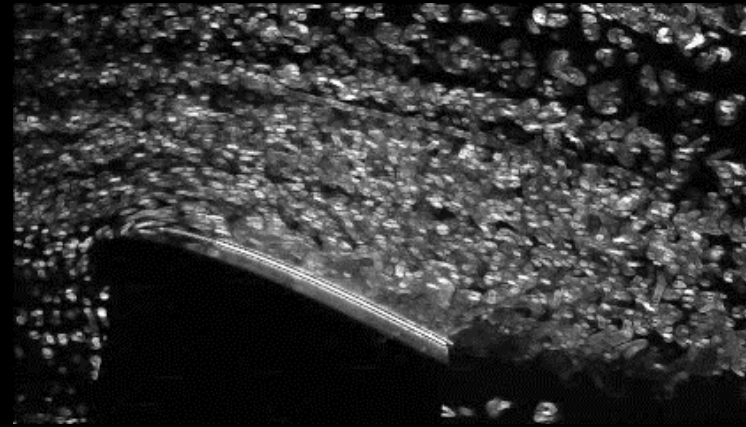


Flow separated near  
the leading edge

Continuous drive



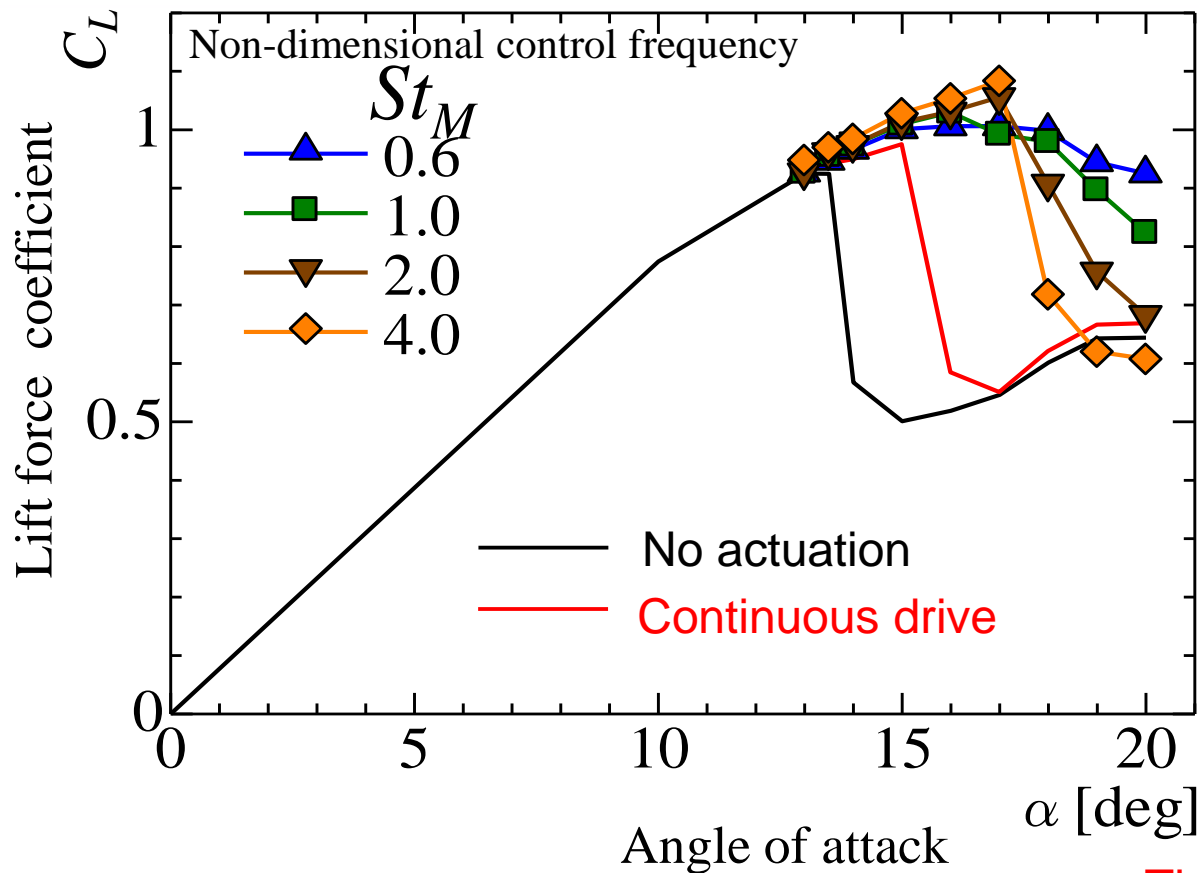
Pulse modulation drive



Leading edge separation is disappeared.

# Result

## Lift coefficient in varying angle of attack



- No control  
 $\alpha_{stall} = 13$  deg
- Continuous drive  
 $\alpha_{stall} = 15$  deg
- Pulse modulation drive  
 $\alpha_{stall} = 17$  deg

The control can delay the stall,  
and increase maximum lift.

# 3rd control example

## Rectangular jet

Turbulent jets are used in a variety of industrial applications such as mixing and cooling/heating.

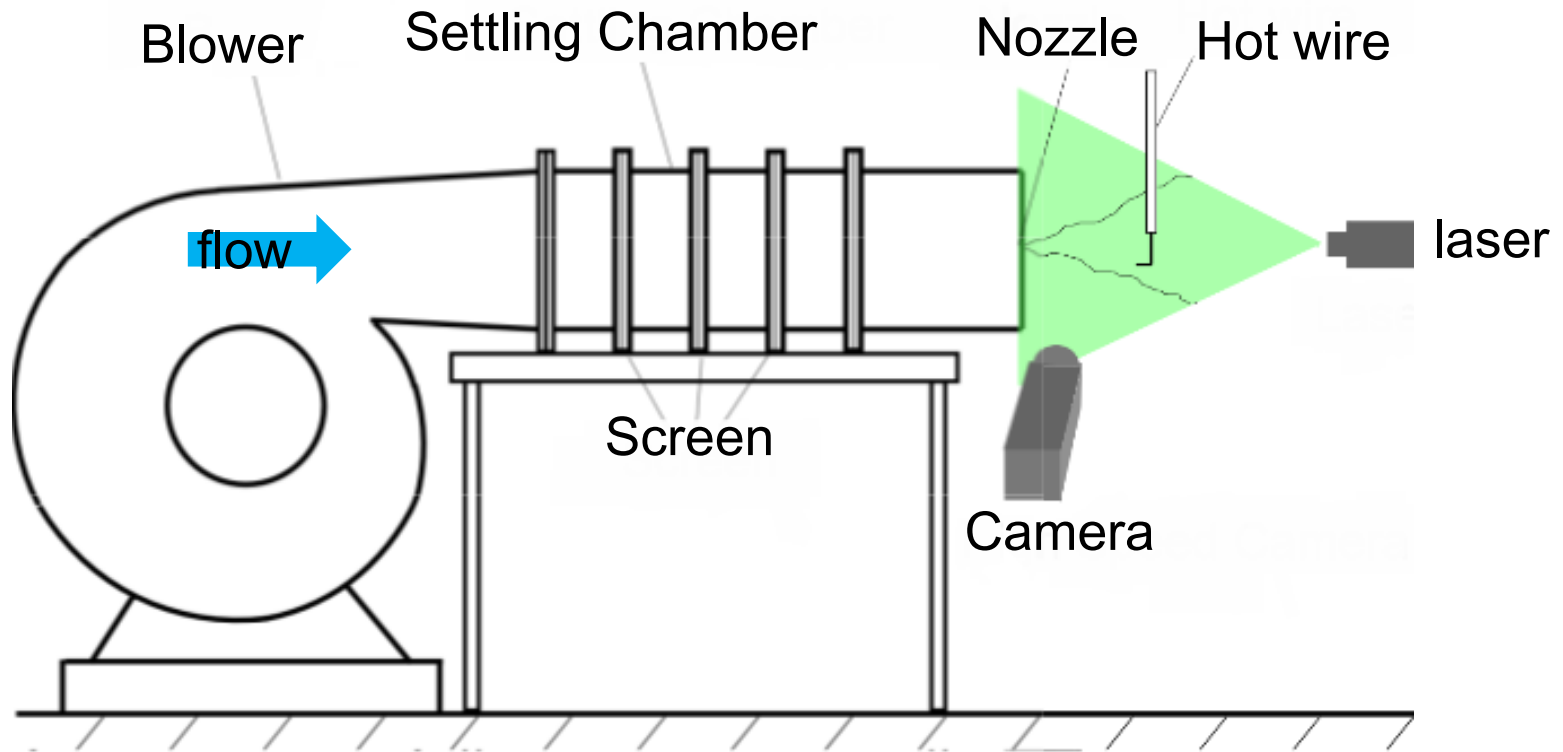


Control of the width of the jet



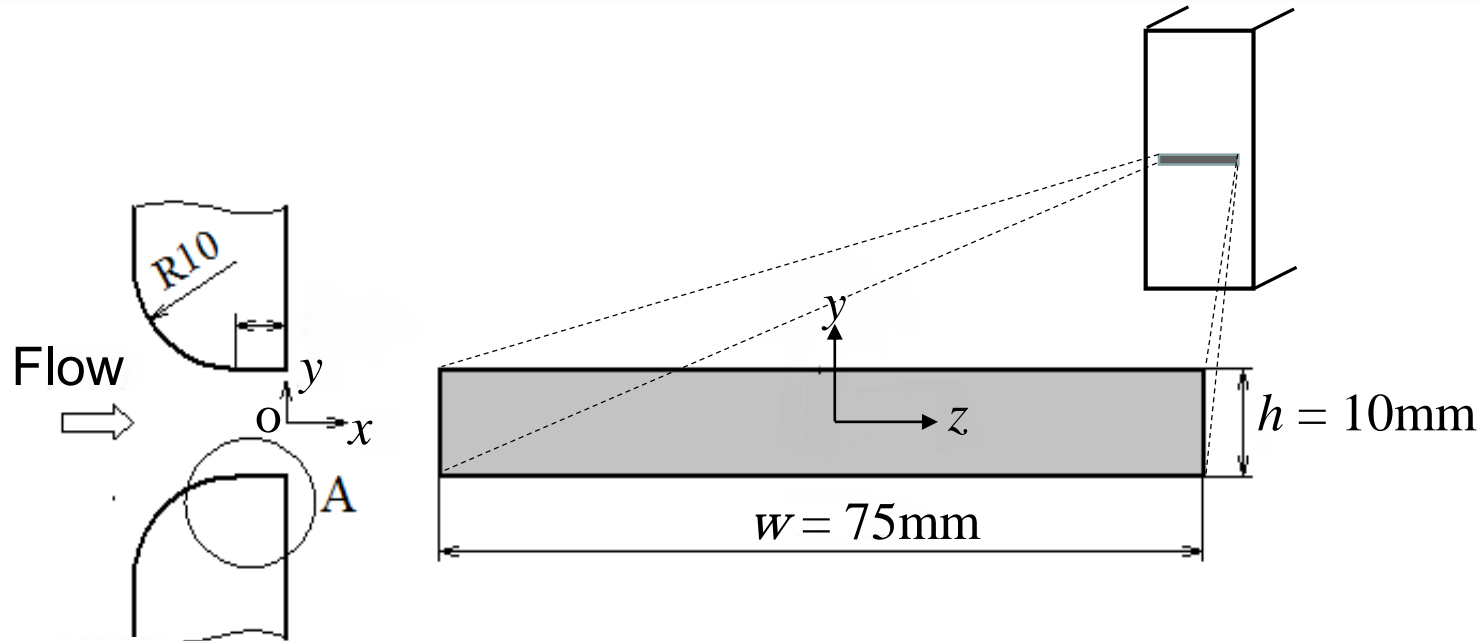
Control of the mixing and cooling/heating

# Experimental Apparatus

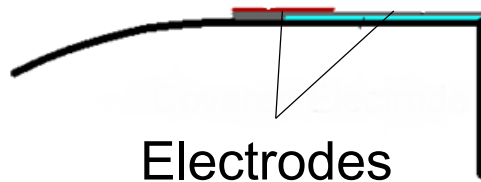


# Experimental setup

## Detail of the nozzle



Detail A



Nozzle exit velocity

$$U_0 = 5\text{m/s}$$

$$Re = U_0 h / \nu = 3,300$$



# Result of control

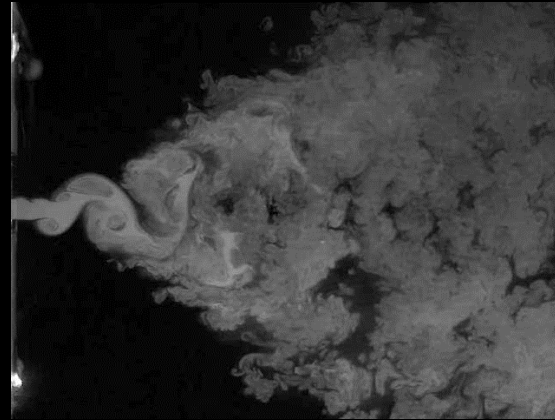
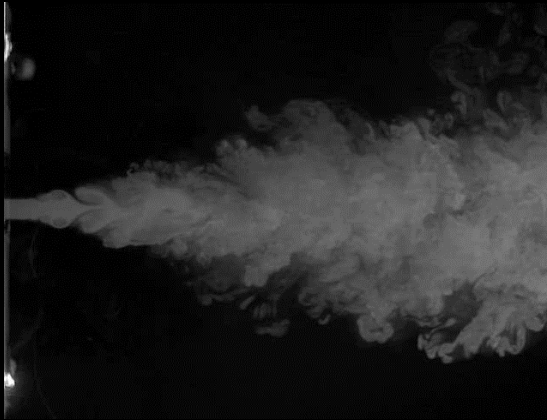
(play speed 1/267)

Base case

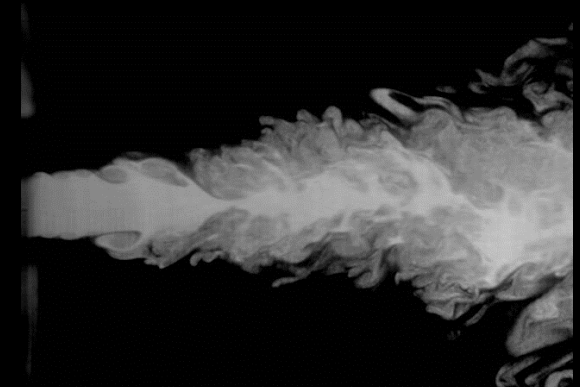
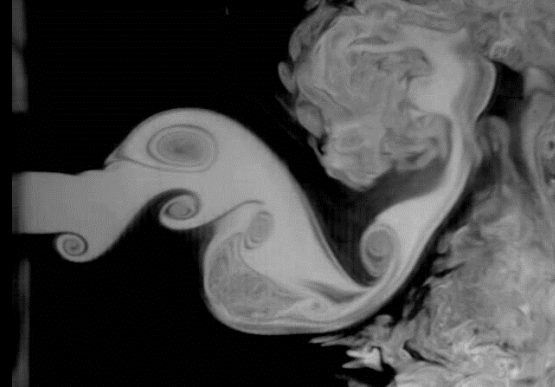
Spread case  
( $f_M=55\text{Hz}$ ,  $Duty=10\%$ ,  $\theta=\pi$ )

Reduced case  
( $f_M=700\text{Hz}$ ,  $Duty=30\%$ ,  $\theta=\pi$ )

Wide view



Close-up view



Large vortex is generated in the up and down, alternately.

Vortex does not become large near the nozzle  
It broke down in a little while.

We can control jet width by DBD plasma actuator.

# Conclusion

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Pulse modulated drive of the DBD plasma actuator is effective for the separation flow control.